Nonlinear Dynamics and Systems Theory, 9(2) (2009) 147-160



State Dependent Generalized Inversion-Based Liapunov Equation for Spacecraft Attitude Control

Abdulrahman H. Bajodah*

Aeronautical Engineering Department, P.O. Box 80204 King Abdulaziz University, Jeddah 21589, Saudi Arabia

Received: June 1, 2007; Revised: October 20, 2008

Abstract: Parametrization of nonunique linear equations solution via generalized inversion is utilized in nonlinear spacecraft control system design. A stable linear time-invariant ordinary differential equation in an attitude deviation norm measure is formed and is evaluated along the trajectories defined by the spacecraft mathematical model, yielding a linear relation in the control variables. Generalized inversion of the relation results in a control law that consists of auxiliary and particular parts. The *null-control vector* in the auxiliary part is designed by solving a state dependent Liapunov equation involving a *perturbed nullprojector* and by utilizing a *damped controls coefficient generalized inverse*, yielding globally uniformly ultimately bounded attitude trajectory tracking errors.

Keywords: spacecraft attitude control; Moore–Penrose controls coefficient generalized inverse; null-control vector; damped controls coefficient generalized inverse; state dependent Liapunov equation; perturbed controls coefficient nullprojection.

Mathematics Subject Classification (2000): 93B52, 93C10, 93C15, 93C35, 93C73, 93D05, 93D15, 93D30.